

On the stability and accuracy of high stiffness rendering in non-backdrivable actuators through series elasticity

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Abstract

This paper addresses the problem of accuracy and coupled stability of stiffness-controlled series elastic actuators, where the motor is modeled as a non-backdrivable velocity source, and the desired value of virtual stiffness is above the physical stiffness of the compliant element. We first demonstrate that, within the mentioned conditions, no linear outer-loop force control action can be applied on the velocity-sourced motor to passify the system. Relaxing the constraint of passivity, we exhaustively search the control design space defined by parametric force and stiffness controllers, expressed in a general lead-lag form, and define a lead-type stiffness compensator that results in acceptable conditions for both coupled stability and accuracy. We also address the effect of a non-ideality in the velocity control loop, such as limited-bandwidth velocity control, and derive relationships between the value of the inner velocity loop time constant and parameters of the stiffness compensator that provide the best performance in terms of both stability and accuracy of haptic display, and test our optimized controller both through numerical simulations and through experiments with the MR-SEA II.

We show that the parameters of a simple outer-loop stiffness compensator can be optimized to result in a stable and accurate display of virtual environments with stiffness values in a large range, that also comprises values of virtual stiffness *higher* than the physical stiffness of the compliant element. A

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requirement for coupled stability is that the actuator is designed such that the minimum value of inertia connected to the compliant actuator load is higher than a control-defined threshold. Finally, we extensively analyze how the minimum value of interaction mass for coupled stability can be minimized through modulation of the stiffness compensator zeros and poles, considering realistic limitations in the velocity control bandwidth of non-backdrivable motors. Our analysis, validated through both numerical simulations and experiments, opens the possibility for alternative approaches to the design of compliant actuators, whereby rendering of high stiffness is possible if the load mass is always higher than a determined threshold.

Keywords: Compliant actuators, haptics, coupled stability, force control

1. Background

2 The possibility of safe physical interaction and successful cooperation with
humans is among the most promising and exciting frontiers of robotics. Several
4 new scenarios such as rehabilitation robotics [1, 2, 3], human augmentation
robotics [4], surgical robotics [5, 6] and haptics have rapidly transitioned from
6 science-fiction, to research laboratories to flourishing industries. In all those
scenarios, a common underlying feature is the need of regulating the physical
8 interaction between a human and a robot.

 In order to achieve this goal, interaction control approaches, as described
10 in the impedance control framework [7], have extensively been implemented,
and their effectiveness demonstrated in several applications requiring physical
12 interaction with humans. In the case of simple impedance control [7], [8], the
controller is defined in impedance causality, and the mechanism is modeled as
14 a pure source of effort variables. Successful implementation of this type of
impedance controller has been demonstrated mostly in the case of robots with
16 negligible intrinsic dynamical properties or whose motion is approximated by
quasi-static movements [1, 9, 3].

18 The problem of accurately regulating interaction becomes more difficult in
the case of manipulators with complex dynamics, with high inertia, and/or with
20 highly non-transparent actuation systems [10]. When robots are intended for
applications requiring substantial assistance to humans during load-intensive
22 tasks, manipulators are indeed not pure effort sources, due to the lack of actu-
ation systems that allow achievement of high force density without substantial
24 increase of task-space dynamic loading. In this case, model-based dynamic
compensation schemes for impedance control can be adopted [11], but often
26 do not fully guarantee accurate interaction control. In general, effectiveness of
model-based schemes is limited by neglect of higher order or nonlinear dynami-
28 cal effects, and by issues related to sensorization and practical limitations in the
capabilities of full state feedback, required for compensation of inertial loads.

30 Force feedback, an approach pursued since the late 1970s [12], can enable
the accurate regulation of interaction also in non-transparent manipulators. In
32 such architectures, an explicit measurement of the force of interaction between
the manipulator and the environment is used to generate a command signal,
34 that ultimately regulates interaction with the environment. Since then, several
forms have been proposed for the force controller, such as proportional control,
36 pure integral control, proportional-integral (PI) control [11], and controllers
with inner motion-control loops ([13, 14] for compliant joints, and [15] for rigid
38 actuators).

1.1. Passivity for force-feedback systems

40 Real-world implementations of interaction control through force-feedback do
not succeed in achieving arbitrary impedance values. In practice, when a con-
42 troller attempts to emulate dynamics that differ significantly from those of the
hardware, the risk of instability increases [16]. The stability limits in force-
44 feedback controlled systems have been approached through the concept of pas-
sivity, a concept adapted from classical theory of electrical networks [16]. It has
46 been proved that when two stable systems with passive impedance port function
are coupled together, the coupled system, that results from the connection of

the two systems, is stable. Instead, if a robot is stable but non-passive, there will be at least a passive environment that, during interaction, will destabilize the controlled system [16, 17]. Proving passivity of a controlled system ensures stability for a wide range of interaction environments that include human dynamics, that are generally modeled as a passive, non linear, first- or second-order system. Also, passivity can be proved for a simplified model, that includes only the controlled system, and do not require detailed knowledge of the environment. Despite the introduced simplicity, this approach allows derivation of strong conclusions on the stability properties of the robot, when coupled to an extremely large and useful set of environments.

Requiring passivity has also drawbacks. Colgate showed that if endpoint force feedback is used to compensate for also the distal mass, the system becomes non-passive [17]. This limitation results from the often unavoidable presence of dynamics between the force sensor and the actuator, that can severely limit the performance of force and impedance controllers. Achievement of global passivity properties for a controller (i.e. passivity for all frequencies) is considered an important requirement for human-interacting robots; yet, at the same time, it is also acknowledged to be quite conservative [18]. It is indeed recognized that the frequency-domain passivity requirement often poses excessive limitations on performance. Examples of approaches violating the frequency-domain passivity requirement without limitations of coupled stability are time-domain passivity controllers [18], also implemented in teleoperation systems [19], and force-feedback controllers for wrist robots[20].

1.2. Inclusion of physical compliance for interaction control

An alternative approach to improving the reflected dynamics of manipulators requires considerable changes to actuators design, as done in Series Elastic Actuators (SEA)[21, 22], where a compliant element is introduced in series between the actuator and the load and its deflection measured. This measurement enables estimation of the interaction forces exchanged between the actuator and the environment, and ultimately of the interaction forces between the human

78 and the robot.

SEAs were originally proposed for their mechanical advantages over stiff
80 actuators, such as shock tolerance and increased power capabilities [21, 23, 24].

In later years, several research groups showed that Series Elastic Actuators
82 can be successfully employed for accurate implementation of interaction control
approaches with actuation systems that could not be modeled as low-impedance
84 effort sources [25, 26, 27]. From a control perspective, the primary advantage of
series elasticity over stiff force feedback is that the compliant force sensor reduces
86 the physical gain of the feedforward path in the force control loop. In this way,
the control gain can be proportionally increased to maintain the overall loop
88 gain of the actuator, resulting in the same stability margins with higher control
gains [22]. SEAs can then display a lower output impedance than the one of the
90 actuator alone, without imposing the same stringent limits on the maximum
reduction of endpoint inertia as is the case for systems with stiff force-feedback.
92 In fact, this scheme allows simultaneous adoption of high-g geared motors and
achievement of low apparent load inertia, since motor inertia is decoupled from
94 the load through the series elastic element.

Though the requirement on minimum reflected inertia for coupled stability
96 does not apply to SEAs, it has been demonstrated that, if inner velocity loops are
introduced in the control architecture, the SEA is not passive if it attempts to
98 regulate a behavior corresponding to a pure spring with elastic constant higher
than that of the physical compliant element [14]. In the following, we will refer
100 to this case as “virtual stiffness control”, in which it is desired to regulate the
force of interaction F_i , in a way that it is proportional to the error between the
102 measured position x and a desired position x_{des} through a constant K_{des} , named
virtual stiffness, so that $F_i = K_{des}(x_{des} - x)$. Although different passivity con-
104 ditions are obtained through alternative controllers types, the presence of inner
velocity loops is often mandatory. This is certainly the case of non-backdrivable
106 actuators which can be more suitably modeled in admittance causality, such
as piezoelectric actuators. For this class of actuators, cascaded force-velocity
108 represents the most direct implementation of an interaction controller. Unfor-

110 Unfortunately, the limit on passivity (achieved only for $K_{des} \leq k_s$) poses a stringent
 112 limitation on the maximum stiffness that can be accurately rendered through
 a compliant actuator that includes a non-backdrivable motor. Practical exam-
 114 ples of such velocity-sourced motors are ultrasonic piezoelectric motors, such
 as the one commercialized by Shinsei and utilized in MR-compatible robotics
 applications [28, 29].

This paper investigates the consequences arising from the violation of the
 116 passivity requirement, when the actuated system is controlled to render a pure
 virtual stiffness with elastic constant higher than the physical spring of the
 118 SEA. The analysis is conducted for non-backdrivable motors that are modeled
 as ideal velocity sources, for which it is possible to derive the parameters of a
 120 stiffness compensator capable of achieving coupled stability with a wide range of
 passive environments. The approach is finally validated through both numerical
 122 simulations and experiments in a 1-DOF test bench.

2. Modeling and problem definition

124 The schematic of a SEA is presented in Fig. 1, in which an actuator drives
 the output mass through a spring-mass-damper system. The compliant actuator
 126 regulates the force of interaction with the environment F_L through measurement
 of x_M and x_L , representing motor and load displacements, respectively. Assum-
 128 ing knowledge of the spring elastic constant k_s , F_L is estimated through Hook
 law: $F_L = k_s(x_L - x_M)$.

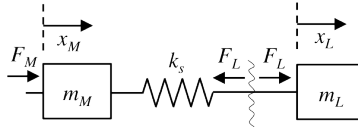


Figure 1: Mechanical schematic of a Series Elastic Actuator. A motor is connected to
 the load through a spring, whose deflection is measured, thus allowing measurement
 of the interaction force F_L .

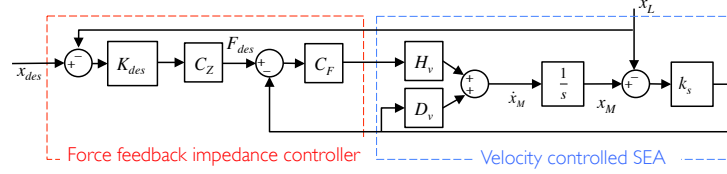


Figure 2: Block diagram of the controlled system. A non-backdrivable motor is controlled to be a velocity source, and an outer loop is closed on the measured force of interaction with the environment. A velocity control disturbance transfer function can be defined to describe the effect of interaction force on error in velocity control. This function will be neglected, assuming negligible admittance for the velocity controlled motor.

130 The goal is to develop a controller which is able to regulate the apparent
impedance $Z_L = \frac{F_L}{\dot{x}_L} = \frac{k_s(x_L - x_M)}{\dot{x}_L}$ of the system at the port of interaction with
132 the environment, (i.e. at coordinate x_L), through only measurement of the
two positions x_L and x_M . In particular, we are interested in the possibility
134 of rendering the behavior of a pure spring, with virtual constant K_{des} higher
than that of the physical spring, k_s . The environment is defined by the mass
136 m_L in the figure, but it could be any linear passive environment, (i.e. any
combination of spring and mass). To this aim, we will investigate whether there
138 are control actions that guarantee passivity of the transfer function Z_L , that
would consequently imply stability of the system when coupled to any passive
140 environment.

3. Cascaded force-velocity control

142 Several control approaches have been proposed for force control of SEAs,
including direct feedback force controllers with feedforward compensation [30],
144 nonlinear compensators capable of reducing the effects of friction and variabil-
ity of interaction dynamics [31] and cascaded linear force-position [32] or force-
146 velocity control [33, 13, 34]. In the specific case of SEAs designed to improve
the interaction control performance of non-backdrivable or high impedance ac-

148 tuators, the latter two controllers are of particular interest, since they effectively
allow the conversion of a force control problem into simpler position or velocity
150 control problems, thereby enabling the implementation of interaction controllers
for non-transparent motors. The cascaded force and velocity control scheme was
152 demonstrated to be passive [34] for a wide range of desired impedance values.
This holds true even in the absence of viscous friction, which is not generally
154 the case of direct force/torque feedback control [35].

Fig. 2 reports a general block diagram of the cascaded force-velocity con-
156 troller for an SEA in the Laplace domain, C_Z and C_F being the transfer func-
tions of the stiffness and force controller, respectively. The inner velocity loop
158 can be modeled by the superposition of two contributions, one describing ve-
locity control performance, and the other describing the degradation of velocity
160 control due to the interaction with the environment:

$$V_M(s) = H_v(s)V_{des}(s) + D_v(s)F_L(s), \quad (1)$$

where H_v is the velocity control closed loop transfer function, in the absence of
162 torque disturbance, and D_v describes the effect of torque disturbance on velocity
control output.

164 For non-backdrivable motors, within their linear range, the effect of veloc-
ity reduction provided by interaction with the environment is negligible. Such
166 simplification is accurate for a class of actuators such as piezoelectric actua-
tors, which have a very small intrinsic admittance both in their unpowered and
168 in their velocity-controlled modes [36]. In order to model this effect, we will
assume perfect disturbance rejection from the velocity controller, and we will
170 assume that the term $D_v(s)F_L(s)$ is negligible compared to the first term in
(1), implying no effect deriving from load torque on the motor velocity control
172 performance.

Under the mentioned assumptions, the velocity-controlled ultrasonic motor
174 has been modeled as an ideal velocity source, as described in the block diagram
shown in Fig. 2, assuming that $D_v = 0$. In order to regulate interaction, an
176 outer loop is then closed on the measurement of the load position x_L , which is

compared to the desired load position, in order to define a desired force value
178 F_{des} , determined as a stiffness force field converging towards the desired position
 x_{des} . In interaction control, it is often desired to regulate mechanical interaction
180 between subjects and the robot by specifying different values of virtual stiffness
 K_{des} , thereby regulating the level of mechanical assistance towards the equilib-
182 rium position provided by the robot. In the following, we will analyze which are
the limits of virtual stiffness values that can be rendered in a stable manner,
184 and design control actions to improve stability and accuracy of stiffness control.

3.1. Limitations of pure stiffness control through cascaded force-velocity control

The most straightforward way to implement a stiffness controller through
the block diagram shown in Fig. 2 is to impose $C_Z = 1$. Correspondingly,
the outer stiffness loop will be commanding elastic desired interaction forces,
displaying a force field proportional to the difference between the desired load
position and the measured load position, with an arbitrary stiffness constant
 K_{des} . In this case, using a proportional controller as force compensator (i.e.
 $C_F = k_{p,f}$), the impedance of the controlled system Z_L can be calculated as:

$$Z_L = \frac{k_s s + (K_{des} k_{p,f} k_s)}{s^2 + (k_{p,f} k_s) s} \quad (2)$$

Analysis of the passivity of the controlled system impedance transfer function
allows determining the coupled stability during interaction with passive environ-
ments. In order for the controlled system to be passive, the impedance transfer
function Z_L needs to be stable and to satisfy the condition $Re(Z_L(\omega)) > 0, \forall \omega \in$
 R . Through symbolic calculation, assuming $s = j\omega$, we can calculate the real
part of the Z_L transfer function, and evaluate conditions for it to be positive,
hence concluding on the passivity of the system. We obtain

$$Re(Z_L(j\omega)) = \frac{k_s k_{p,f} (k_s - K_{des})}{\omega^2 + (k_{p,f}^2 k_s^2)}, \quad (3)$$

186 which is clearly positive only for $k_s > K_{des}$, for all values of the force compen-
sator gain. The same result can be derived for a force compensator that includes
188 a proportional-integral controller, as the one proposed for passive interaction

control of SEAs [14]. Hence, the system is not passive if it is commanded to
 190 render a virtual spring with stiffness higher than that of the physical spring.

4. Coupled stability and performance during display of high-stiffness 192 environments

The analysis of the Bode plot of the impedance transfer function of the
 194 controlled system is shown in Fig. 3. It can be seen that when $K_{des} > k_s$, the
 impedance transfer function approximates a pure spring at low frequencies, and
 196 becomes a non-passive transfer function (with phase lower than -90 deg) above
 a certain frequency value, that can be modulated through action on the force
 198 feedback compensator gain.

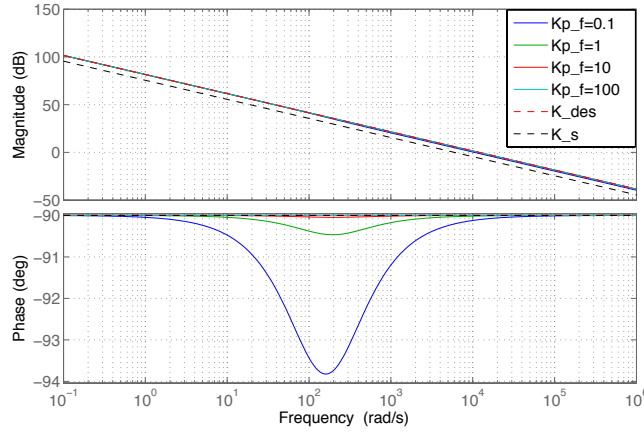


Figure 3: Bode diagram of the apparent impedance of the system, when controlled
 through the proportional impedance compensator, $F_{des} = -K_{des}(x_L - x_{des})$, with a
 proportional force feedback compensator $C_F = k_{p,f}$.

In order to guarantee stability during rendering of a high-impedance envi-
 200 ronment, we relax the constraints on the specific form of force and stiffness
 compensators, and conduct an analysis in which we will assume them to be
 202 generic, causal compensators C_F (for the force compensator), and $K_{des}C_Z$ (for
 the stiffness compensator), and consider a causal rational polynomial expres-

204 sion for the velocity control transfer function H_v . The virtual impedance Z_L
 rendered at the load side, is given by:

$$Z_L(s) = \frac{k_s}{s} \frac{s + K_{des} C_F H_v C_Z}{s + k_s C_F H_v} = \frac{k_s}{s} K(s) \quad (4)$$

206 Analysis of (4) provides significant insight on the behavior of the system. In
 particular, the first term defines the physical properties of the system reflected
 208 to the output: a pure spring with stiffness k_s . In contrast, the second term,
 labeled as $K(s)$, is a transfer function that describes how the physical stiffness
 210 is modulated to be displayed to the output. If $K(s) = 1$, the system displays
 a stiffness exactly equal to that of the physical spring. Modulations in the
 212 amplitude of $K(s)$ allow regulation of different stiffness values. If the phase of
 $K(s)$ is positive, this will add damping to the system, whereas if the phase
 214 of $K(s)$ is negative, this will contribute to further decreasing the phase of the
 impedance transfer function Z_L below its value of -90 deg dictated by the pure
 216 integral action of the spring. In particular, it should be noted that, unlike the
 case of servo controls, the transfer function of force and impedance compensators
 218 do not enter linearly into the open-loop transfer function $Z_L Y_e$ (that determines
 stability of the system when coupled to the environment with admittance Y_e),
 220 nor into the transfer function Z_L (that determines performance of the haptic
 display). This limitation is, in general, inherent to interaction controllers [10]
 222 that operate without a specific model of the environment. This makes design
 of a controller aimed at regulating interaction with an unknown environment a
 224 difficult task that cannot be addressed with traditional methods borrowed from
 the servo controllers literature.

226 It is useful to analyze how the conditions for accuracy and passivity translate
 into requirements for the $K(s)$ transfer function. For pure stiffness control, it is
 228 desired to render a pure spring, with stiffness K_{des} . In this case, the performance
 condition is:

$$Z_L(\omega) = \frac{K_{des}}{j\omega}, \omega \in \Omega_c, \quad (5)$$

where Ω_c is a range of frequencies of interest, for which it is desired to accurately regulate interaction. The condition for passivity of the haptic display dictates that the apparent stiffness fraction $K(s)$ needs to satisfy the following relation, $\forall \omega$:

$$0 < \arg K(\omega) < \pi \quad (6)$$

We now evaluate how the choice of compensators C_F and C_Z allows fulfillment of coupled stability during interaction with the environment. We will focus on the case $K_{des} > k_s$, which is a challenging condition for which to ensure coupled stability. In fact, it can be intuitively demonstrated that no choice of causal, finite-magnitude compensators C_F and C_Z allows fulfillment of the passivity requirement with the cascaded force-velocity control scheme. In fact, if C_F and C_Z have finite magnitude response at all frequencies, $\lim_{s \rightarrow \infty} K(s) = 1$, which implies that at high frequencies, the apparent stiffness of the system reduces to the physical stiffness of the series elastic element. However, for the case in which the desired stiffness is higher than the physical stiffness of the spring (i.e. $K_{des} > k_s$), the modulus of the stiffness transfer function $|K|$, within the range of frequencies Ω_c will be such that $|K(\omega_1)| > |K(\omega)|, \forall \omega_1 \in \Omega_c, \omega > \max(\Omega_c)$. Hence, for $K_{des} > k_s$, the imposition of the performance requirement implies that $K(s)$ is required to have a flat, zero-phase region at low frequencies, with amplitude $K_{des}/k_s > 1$, and a second flat, zero-phase region (above the frequency band Ω_c where performance is specified), of unitary amplitude. Due to a fundamental property of Bode plots, the two zero-phase regions (higher amplitude at lower frequencies and lower amplitude at higher frequencies) can not be simultaneously present in a system, without the phase diagram having at least one point with negative phase. It is thus not possible, with the reported control approach, to render accurate, zero damping, haptic display, with stiffness $K_{des} > k_s$, respecting the passivity requirement. The mentioned limitation has profound consequences: from the analysis of the closed-loop transfer function $Z_L Y_e$, it can be demonstrated that, subject to proportional force control $C_F = k_{p,f}$ and to the implementation of pure stiffness control ($C_Z = 1$),

the system is unstable when coupled to any environment consisting of a mass
 256 $m_e > 0$ (i.e. the worst possible coupled stability result).

4.1. Role of force compensator on coupled stability

Not every choice of force and stiffness compensators C_F and C_Z implies coupled instability with every value of environment mass. We will demonstrate this through two separate analyses that consider different choices of compensators C_F and C_Z . Let us first assume the following structure for the force compensator:

$$C_F = k_{p,f} \frac{s + z_F}{s + p_F} \quad (7)$$

Through this expression, the force compensator is expressed by a transfer function with one zero and one pole that can take the form of a lead compensator (for $z_F < p_F$), a lag compensator (for $z_F > p_F$), a proportional controller $p_F = z_F$, a proportional-integral action (for $p_F \rightarrow 0$) and a causal derivative action (for $z_F \rightarrow 0$). For this controller, the impedance transfer function Z_L^f obtained under the effect of the force compensator is:

$$Z_L^f(s) = \frac{k_s s^2 + (k_s p_f + K_{des} k_{p,f} k_s) s + K_{des} k_{p,f} k_s z_f}{s^3 + (p_f + k_{p,f} k_s) s^2 + (k_{p,f} k_s z_f) s}, \quad (8)$$

258 which, in agreement with the general demonstration presented in section 4, is
 not passive for any choice of the control parameters $k_{p,f}$, p_f and z_f , when
 260 $K_{des} > k_s$. It is interesting to observe the coupled instability of the system,
 when it is interacting with its most destabilizing environment. It can be veri-
 262 fied that for this stiffness controlled system, the most destabilizing environment
 is a pure mass, i.e. the admittance $Y_e = \frac{1}{m_e s}$. In this condition, the closed-loop
 264 system is unstable for every value of interacting load mass $m_e > 0$, for every
 value of the controller parameters. In fact, through calculation of the char-
 266 acteristic polynomial $CLP = 1 + Z_L^f Y_e$ and application of the Routh-Hurwitz
 stability criterion to the coefficients of its numerator, it is possible to simplify
 268 the resulting system of inequalities to the simple expression: $K_{des} \leq k_s$.

This demonstrates that coupled stability can be obtained only for values
 270 of desired stiffness less than or equal to the stiffness of the physical spring,

regardless of the specific choice of the force feedback compensator (i.e. a lead
 272 or lag compensator, PI controller or compensator with a derivative action), for
 every value of interacting mass.

274 4.2. Coupled stability through stiffness compensator

The simplest method to guarantee stability of the system is through an ex-
 276 plicit damping action by the controller, that can be expressed in the form of
 a velocity-dependent action of the stiffness compensator C_Z . This can be ob-
 278 tained, for example, by employing proportional force feedback, and considering
 the following form for the stiffness controller: $C_Z = 1 + \frac{B_{des}}{K_{des}}s$. In this case, the
 280 resulting system can be made passive for certain values of desired stiffness K_{des} .
 In fact, the impedance transfer function Z_L^b , under the action of the damping
 282 action is:

$$Z_L^b = \frac{(k_s + B_{des}k_{p,f}k_s)s + K_{des}k_{p,f}k_s}{s^2 + k_{p,f}k_s s}, \quad (9)$$

which is passive also when $K_{des} > k_s$, if the controller is introducing an amount
 of damping higher than the threshold B^* , defined as

$$B^* = \frac{K_{des} - k_s}{k_{p,f}k_s}. \quad (10)$$

Some practical limitations arise from the adoption of this controller. First, if
 284 no direct measurement of velocity is assumed, the anti-causal damping action
 cannot be practically implemented in real systems. Practically, the damping
 286 term will be implemented as a causal derivative action, which behaves as a
 pure differentiator¹ at low frequencies, and reduces to a proportional gain at
 288 higher frequencies, as provided by the causal differentiator transfer function
 $G = \frac{Ns}{s+N}$. Introducing the non-ideal differentiator transfer function in the
 290 impedance compensator, we then lose again the possibility of proving passivity
 for the impedance-controlled system for every $K_{des} > k_s$. Indeed, the effect

¹We are neglecting the effect of the required low-pass filtering and delay that is needed to
 differentiate digital position signals deriving from encoders, that are affected by quantization.

292 of a non-ideal compensation term is significant. Because of the introduction
of the damping B_{des} , it can be shown that for every finite value of N , coupled
294 instability will occur for every value of environment mass $m_e > 0$, i.e. the worst-
possible coupled stability result. This issue can be mitigated by introducing
296 damping in the system in excess, compared to what is required in theory in
(10). However, this solution would reduce the accuracy of haptic display also
298 at low frequencies, where in theory no damping action would really be needed
for coupled stability. In order to address the limitations of this pure damping
300 action, we will then investigate the effect of the introduction of lead and lag in
the stiffness compensator, by defining it in the form²:

$$C_Z = \frac{p}{z} \frac{s + z}{s + p} \quad (11)$$

In this case, the resulting impedance transfer function, under the effect of the
stiffness compensator, Z_L^k is:

$$Z_L^k(s) = \frac{k_s z s^2 + (k_s p z + K_{des} k_{p,f} k_s p) s + K_{des} k_{p,f} k_s p z}{z s^3 + (p z + k_{p,f} k_s z) s^2 + (k_{p,f} k_s p z) s} \quad (12)$$

302 Again, the introduction of an amplitude-bounded causal compensator C_Z ,
acting on the error between measured and desired load position, results in an
304 impedance transfer function that is passive only if $K_{des} \leq k_s$. However, an
analysis of coupled stability of this controller provides results that allow for
306 margins in controller design. In fact, through evaluation of the closed loop
polynomial during interaction with the most destabilizing environment (again,
308 a pure mass), and application of the Routh-Hurwitz stability criterion, it can be
seen that coupled stability can be obtained for a wide range of interacting mass
310 values, that can be modified by a proper choice of controller gains. Reduction of
the Routh-Hurwitz determinants provides the following conditions for coupled
312 stability ($m_e > 0$):

²The symbolic form of the compensator C_Z is similar to the one chosen for C_F , but an
amplitude normalization factor $\frac{p}{z}$ has been introduced, so as to guarantee that the stiffness
displayed at low frequencies matches the desired stiffness K_{des} .

$$\begin{aligned}
& 1) \quad m_e < m^* \text{ AND } p < \frac{k_s z}{K_{des}} \text{ AND } \left[k_{p,f} \neq \frac{K_{des} z - k_s z}{K_{des} k_s} \right] \\
& 2) \quad m_e > m^* \text{ AND } p > \frac{K_{des} k_{p,f} k_s z}{K_{des} k_{p,f} k_s - K_{des} z + k_s z} \quad (13) \\
& \text{with } m^* = \frac{k_s (K_{des} k_{p,f} + z) (K_{des} p - k_s z)}{z (k_{p,f} k_s + p) [p z k_s + K_{des} (k_{p,f} k_s (p - z) - p z)]}
\end{aligned}$$

showing that coupled stability can be obtained as a function of a threshold load mass value m^* . Though this controller cannot be proved stable with arbitrary values of environment mass m_e , it allows modulation of the value of m^* to ensure that the value of admissible environment mass m_e is always within the stability range defined by (13). In particular, the value of m^* can be reduced to guarantee that the second inequality shown in (13) is respected in all operating conditions. Through this control design method, it is possible to guarantee that the system is stable for values of interacting mass higher than a certain threshold.

This approach can be surprisingly useful, since in most SEA designs, the minimum value of interacting mass m_e is never equal to zero due to inherent design constraints (e.g. a linear SEA will have a load plate that will be used to connect to a load; a rotary SEA will be necessarily connected to some output link, whose inertia adds to the load inertia). Moreover, for applications that involve wearable human-interacting robots, such as rehabilitation or human augmentation, the minimum value of mass connected to the load is often significant. For example, the mass might consist of the inertia of the hand for a wrist exoskeleton, or of the inertia of the lower leg for a knee exoskeleton. Using a constant gain controller, the system can be made stable with masses whose value can be seen in Fig. 4, parameterized in terms of force feedback gain and desired stiffness value. It can be seen that this control approach allows achievement of coupled stability with a limited set of environments, that can be regulated through control parameters, in a way that the range of admissible values for coupled stability is well contained in the range of interacting environments that can be considered in a given application.

338 We now investigate which values of minimum and maximum interacting mass
 m^* that guarantee coupled stability can be obtained with feasible combinations
340 of control parameters. In particular, we conduct this analysis by assuming
a fixed value for the force compensator control gain, and investigate the de-
342 pendency of m^* on the characteristics of the impedance compensator gain, for
different values of desired stiffness.

344 The results of this analysis are shown in Fig. 4, where the regions of com-
pensator values resulting in minimum and maximum values of interacting mass
346 that guarantee coupled stability are contoured by dashed lines. In particular, it
can be seen that pure stiffness control $p = z$ would result in coupled instability
348 for every value of interacting mass (dark red region interposed between the two
colormap regions where stability is possible for some values of mass).

350 From the coupled stability standpoint, it would seem that the best choice of
 C_Z could either be that of a lead-compensator, with one pole and one zero in
352 the high frequency range, or that of a lag compensator, with low-frequency pole
and zero. In fact, in the first case ($z < p$), coupled stability is achieved for values
354 of environment mass higher than a threshold, that can be decreased through
control gains (i.e. moving p towards higher frequencies and increasing the force-
356 feedback compensator gain $k_{p,f}$). In the second case ($p < z$), coupled stability
is achieved for values of environment mass lower than a threshold, which can be
358 again increased through control design (i.e. moving z towards lower frequencies
and increasing the force-feedback compensator gain).

360 Stability is not the only requirement in the design of interaction controllers.
Although the stiffness compensator C_Z can be designed to have unitary magni-
362 tude and zero phase at low frequencies (resulting in an accurate, pure-stiffness
interaction behavior at those frequencies), controller parameters can influence
364 the upper limit of this frequency region. To this aim, we define a measure of
interaction controller accuracy, as described in [10], and include this measure-
366 ment within an optimization framework. The framework would optimize the
combination of control parameters so as to maximize both performance and
368 stability, that for this specific controller could be defined by the range of masses

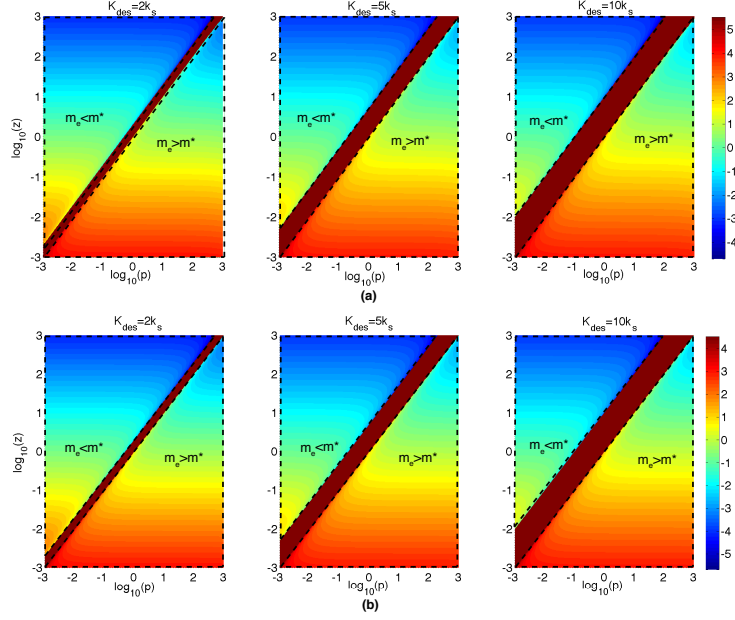


Figure 4: Logarithmic color maps showing values of threshold environment mass m^* , for coupled stability during stiffness control at different values of K_{des} (columns), for different values of force compensator gains, $k_{p,f} = 1$ (a), and $k_{p,f} = 10$ (b). Use of lag compensators ($p < z$ - upper-left triangular section of the $p - z$ plane) result in coupled stability for values of interacting mass $m_e < m^*$, while lead compensators ($p > z$ - lower-right triangular section of the $p - z$ plane), result in coupled stability for values of interacting mass $m_e > m^*$. The region where coupled stability is not guaranteed for any value of interacting mass is separated by the dashed line-contoured regions and is reported in dark red.

that result in a stable interaction.

Under the “ideal non-backdriveability, ideal velocity control” hypothesis stated so far, the system has been simplified to the point that it lends itself to an intuitive, straightforward and informative performance analysis. In fact, from the analysis of the $K(s)$ transfer function, it can be seen that the effect of the impedance compensator is to introduce a phase lead or lag starting at the frequency where the lowest frequency pole or zero of the impedance compensator is located. This can be seen also from the analysis of the impedance transfer function, parameterized as a function of the specific value of compensator gains

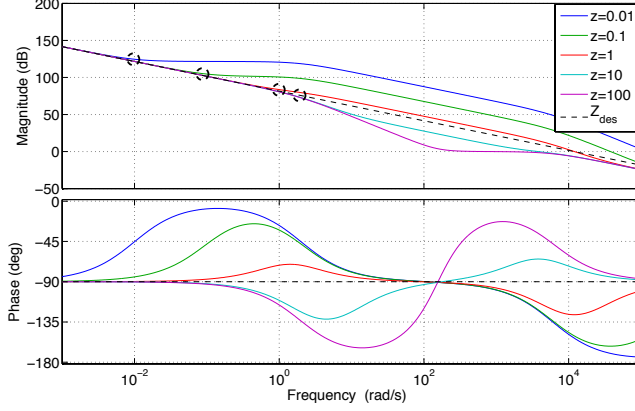


Figure 5: Resulting impedance transfer function Z_L^k , when $p = 1$ and for different values of z , for a unitary force feedback compensator gain and a desired stiffness $K_{des} = 2k_s$. Circled points are those in which the magnitude of the modulated stiffness transfer function $K(s)$ is attenuated by 3 dB. The frequencies in which this situation occurs are in correspondence with the minimum pole or zero of the impedance compensator.

(that change from a lead compensator to a lag compensator). The analysis can be conducted by keeping the pole location constant, and observing the effect on the resulting impedance transfer function, as a function of the zero location, as reported in Fig. 5. It can be seen that accurate impedance rendering is obtained for $\omega \in (-\infty, \min(p, z)]$, i.e. up to a frequency that depends on the minimum value zero or pole introduced in the compensator. A very simple performance measure can be introduced for this system, as given by the two-dimensional function acc , defined as:

$$\text{acc} = \min(p, z), \quad (14)$$

370 with its colormap reported in Fig. 6.

From an intersection analysis between the two plots, it is now clear that
 372 an acceptable compromise between stability and performance is provided only
 by a control action of a lead-compensator type (i.e. $p > z$). In fact, this
 374 controller allows achievement of accurate interaction up to a frequency $\omega_c = z$,
 and guarantees coupled stability for interaction masses higher than a certain
 376 threshold, that can be reduced through control action to values that are largely

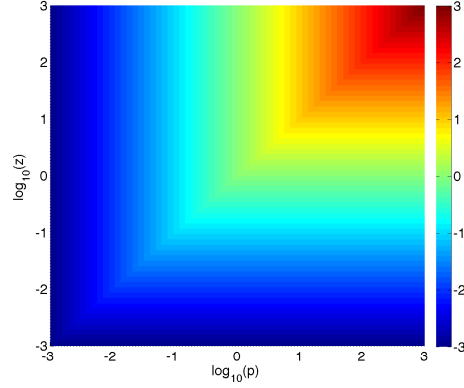


Figure 6: Measure of accuracy for the interaction controller implemented through the impedance compensator C_Z , determined as $\max(p, z)$, in a logarithmic scale. The plotted function represents a quarter of a square pyramid (the lower-left quarter of the pyramid, from the top view shown in the colormap), cut through two orthogonal planes that contain the pyramid axis and one of the axes defined by the basis edge, respectively. The pyramid vertex, i.e. the point $p = p_{MAX}$, $z = p$, corresponds to the point with highest accuracy.

lower than the realistic values associated with a given application.

378 In these conditions, it can be verified by inspection of Eq. (13) that the
value of minimum mass required for a stable interaction is linear with the value
380 of virtual stiffness. This means that, once control gains have been tuned, the
higher the value of desired stiffness K_{des} , the higher the minimum load mass in
382 order to achieve coupled stability.

4.3. Effect of a non ideal inner velocity loop

We now address the effect of model non-idealities, introducing a high-frequency attenuation in velocity control performance, modeling the velocity control transfer function H_v as a low-pass filter $\frac{1}{s\tau+1}$. We describe the effect of τ on both the accuracy and performance of interaction control. First, we analyze the effect of the inner velocity loop transfer function on the selection of the optimal controller gains that maximize stability and evaluate the effect on the haptic display accuracy. For every set of controller gain values, the minimum value of interacting mass for coupled stability is computed through a root locus analysis, analyzing

the interaction with the most destabilizing environment (a pure mass). This can be done by calculating the minimum value of m_e that results in the admittance environment transfer function $Y_e = \frac{1}{m_e s}$ such that the polynomial CLP , that describes the stability of the coupled system, with

$$CLP = 1 + Z_L Y_e \quad (15)$$

has roots in left half complex plane. Also, for every set of controller gain values, a measure of impedance control accuracy is introduced as in [10], as the reciprocal of the cost function, calculated in a logarithmically-spaced set of frequencies of interest $\Omega = \{\omega_1, \omega_2, \dots, \omega_n\}$, $\omega_1 = 10^{-3}$, $\omega_n = 10^3$:

$$C(\omega) = \sum_{i=1}^n |\log |Z_L(j\omega_i)| - \log |Z_{des}(j\omega_i)||, \quad (16)$$

that compares the amplitude of the resulting impedance transfer function Z_L to the desired impedance Z_{des} . The analysis is reported in Fig. 7 for specific values of K_{des} and τ , varying controller gains. The analysis shows that the region of controller gains that maximizes accuracy is, in agreement with the results obtained in the ideal velocity control case, the upper-right corner of the p - z plane, i.e. the region that maximizes p and z , with $p > z$. The introduction of the non-ideal velocity controller slightly modifies the optimality conditions for coupled stability. In fact, minimization of the environment mass for coupled stability is obtained when the compensator zeros/poles are in a lower frequency range. The specific value of the upper limit of this frequency range will depend on the value of the velocity loop transfer function time constant.

To further investigate the issue of optimal zero-pole location in the impedance compensator C_z , we addressed both stability and accuracy, and evaluated how such properties vary as a function of inner velocity loop time constant τ . Once the controller parameters p_{opt}^τ and z_{opt}^τ that yield the minimum value of interacting mass for coupled stability are determined for every value of τ , the accuracy measure $1/C_{opt}^\tau$ of the impedance-controlled system is calculated, and reported (see Fig. 8; labeled as Opt_τ). Similar analysis was conducted for the ideal compensator gains obtained for the ideal velocity loop case (labeled as Opt),

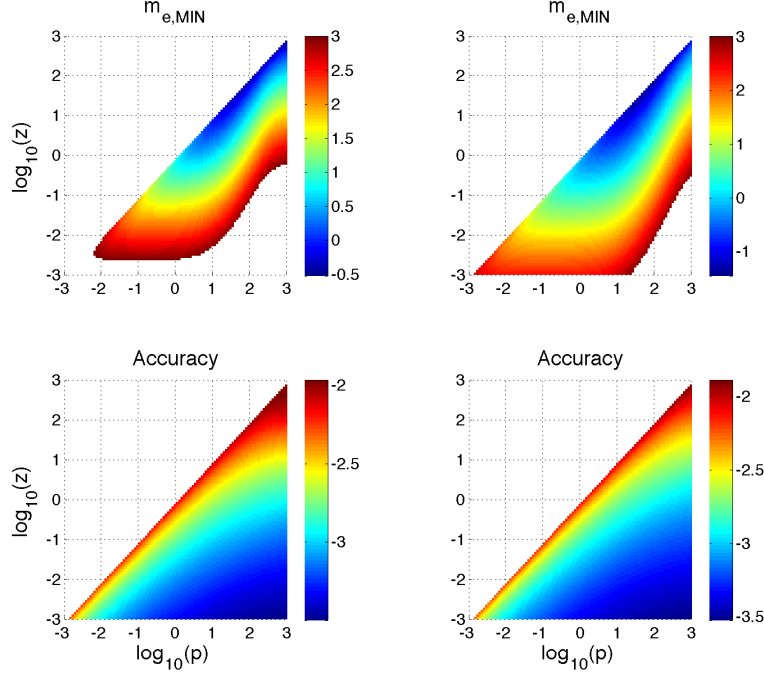


Figure 7: Effect of stiffness compensator parameters on the minimum value of interacting mass for coupled stability ($m_{e,MIN}$, shown in the top plots) and on the accuracy of the impedance transfer function, defined as $1/C$ - lower plots, for frequencies logarithmically spaced between 10^{-3} rad/s and 10^3 rad/s. An inner velocity loop modeled as a first-order low-pass filter with time constant $\tau = 0.01$ is considered in this analysis, and the value of the force-feedback compensator gain is set to 1 (left) and 10 (right).

and for the cases in which the minimum damping B^* , as expressed in eq. (10), is introduced in the compensator (labeled as B^*), or increased damping equal to $2B^*$ was introduced, labeled as $2B^*$. The analysis is graphically reported in Fig. 8, and enables description of the achievable trade-off between stability and accuracy of high-stiffness impedance control in compliant robots with non-backdrivable actuators, controlled as velocity sources, as a function of the responsiveness of the inner velocity loop.

First, it can be seen that the accuracy and coupled stability requirements are

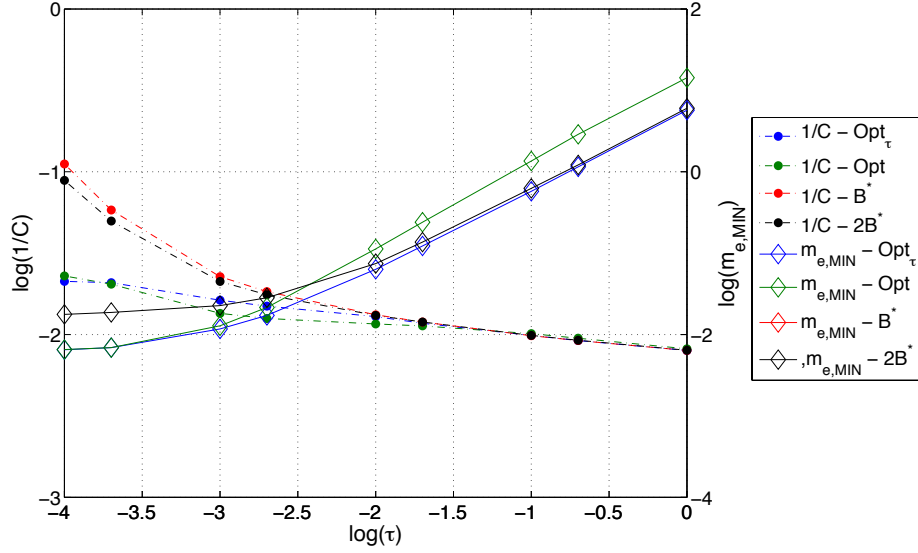


Figure 8: Effect of different inner velocity loop time constant τ on the accuracy and stability of the impedance controlled compliant actuator. Circle-marked dashed lines represent the performance measure in the left logarithmic scale, while diamond-marked continuous lines represent the stability measure, in the right logarithmic scale.

both negatively influenced by slower inner velocity loops. For all considered controllers, a faster inner velocity loop is able to simultaneously provide increased accuracy within the considered frequency range, and lower values of minimum interacting mass for coupled stability (increased stability performance).

A full quantitative comparison between the designed compensator and the system controlled through a pure damping action where the minimum damping B^* is injected in the system is not possible, due to the fact that such a controller results in an unstable system, for all values of interacting mass $m_e > 0$ (due to consideration of frequency-bounded controller derivative action). For this reason, it is not possible to define a measure of stability of this controller.

From a comparison between the results achievable with the two different controllers (i.e. Opt vs. Opt $_{\tau}$), it can be seen that the solution found through separate optimization for different values of τ (Opt $_{\tau}$) provides the best coupled

424 stability results for every value of τ , compared to the controller designed for the
ideal velocity loop case (Opt). The amount of reduction in values of minimum
426 mass that provides coupled stability (Δm_e) allowed by the Opt_τ controller is
very small at low values of τ , ($\frac{\Delta m_e}{m_e} \leq 5\%, \tau \leq 0.01$), and becomes more signifi-
428 cant while τ increases, settling to $\Delta m_e = m_e$, for the case in which $\tau = 1$.

It can be seen that at low values of velocity control time constant ($\tau < 0.01$),
430 coupled stability can be achieved with very low values of interacting mass, at
the expense of haptic display accuracy (the minimum value of interaction mass
432 is reduced in the Opt and Opt_τ controllers to less than half of what is allowed
with the $2B^*$ controller, but accuracy is reduced by a similar amount). This
434 results from the fact that the range in which control parameters p and z are
sought is $[10^{-3} \ 10^3]$ rad/s, representing realistic hardware limitations in real-
436 time control applications. In general, it is shown that the performance obtained
through the controller determined for the ideal velocity loop case (Opt) loses
438 its optimality when the region in the frequency domain in which poles and zeros
of the outer loop compensator are sought ($[10^{-3} \ 10^3]$ rad/s) overlaps with the
440 region at which the velocity control performance degrades.

5. Model validation

442 The feasibility of the developed controller is validated through numerical
non-linear modeling and simulations, as well as through experiments in a com-
444 pliant actuator (MR-SEA II) that includes a non-backdrivable traveling-wave
ultrasonic motor, a cable transmission and two pre-loaded extension springs.

5.1. System description and modeling

The 1 DOF actuator prototype (MR-SEA II) was developed for an MR-
448 compatible haptics application [36] and comprises a rotary ultrasonic piezoce-
ramic motor (5W Shinsei ESR60-E3N), a threaded pulley on the motor shaft, a
450 cable transmission, pre-extended phosphor bronze extension springs, a custom-
designed Delrin carriage and a linear ceramic balls bearing, supporting the

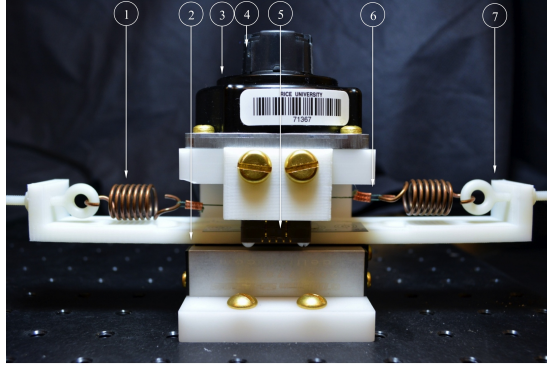


Figure 9: MR-SEA II. (1) phosphor bronze extension spring, (2) ceramic linear ball bearings, (3) piezoceramic motor, (4) rotary optical encoder, used to measure x_M , (5) linear encoder, used to measure x_L , (6) cable transmission, (7) slider, with plastic eyebolts.

452 carriage. As shown in the companion design paper [36], the motor is non-
backdrivable and when controlled through its own factory-tuned velocity con-
454 troller can be modeled as a low admittance velocity source, and compensated to
regulate velocity linearly (i.e. without amplitude dependence, except for its low
456 velocity region) within its operating frequency and amplitude range (see Table
1). A numerical model is implemented in Simulink (the Mathworks Inc.), and
458 run with an ODE 45 solver at a fixed step of 1 ms, to validate the theoretical
analysis assessing the effects of model non-linearities on the closed-loop stabil-
460 ity of the system. In the model, the actuator is controlled to display a stiffness
 K_{des} through the action of the impedance compensator C_Z acting on the cas-
462 caded force-velocity controller, with desired position $x_L = 0$. This controller
action responds to a load force perturbation, modeled as a step with amplitude
464 of 4 N (20% of the actuator force capabilities). The case of force perturbation
when the compliant actuator is interacting with an environmental mass m_e is
466 chosen because it closely matches the task conducted during the experimental
characterization, and because we have demonstrated that it corresponds to the

Table 1: MR-SEA performance

Property	Value
Maximum continuous force	20 N
Spring stiffness	3.8 N/mm
Maximum velocity	0.1 m/s
Velocity control frequency range	0 - 10 Hz
Motor encoder quantization	$1.3 \cdot 10^{-2}$ mm
Load encoder quantization	$1 \cdot 10^{-2}$ mm
Load mass	100 g

most destabilizing environment for the stiffness-controlled actuator³. To separate the effects of model non-idealities, the red blocks in Fig. 10 are introduced sequentially in the analysis.

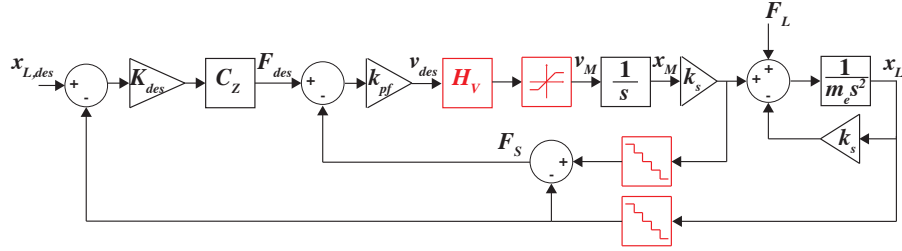


Figure 10: Block diagram of the numerical model used to validate the theoretical analysis of coupled stability. The blocks describing the plant non-idealities are shown in red, and introduced sequentially in the numerical analysis.

³The choice of a particular form of perturbation (e.g. load force perturbation, load velocity perturbation, desired load position) does not change the coupled stability properties of linear systems, since the poles of the closed-loop transfer function are independent of the selected form of perturbation

5.1.1. Effect of plant non-idealities through numerical simulations

After checking through fixed-step numerical integration that our theoretical analysis accurately predicts the value of interaction mass resulting in coupled stability, for the lead-lag compensator, and that no proportional stiffness controller can obtain coupled stability for any value of environment mass $m_e > 0$, when $K_{des} > k_s$, we introduce plant non linearities in the numerical model. Firstly, we introduce the saturation block that models the amplitude-dependent limit of velocity control. We determine through a linear search procedure the values of controller parameters that result in coupled stability with the lowest-possible environment mass $m_{e,MIN}^{sat}$. We then analyze what is the effect of the inclusion of two other non-idealities, i.e. the H_V transfer function (resulting in a minimum value of interaction mass m_{th}^{freq} for coupled stability) and, finally, of encoders quantization (resulting in the value of interaction mass $m_{e,MIN}^q$ that provides coupled stability). The results of this analysis are reported in Table 2. When the value of environment mass reduces to only the mass of the slider, it is theoretically possible to achieve coupled stability for values of K_{des} up to $1.75k_s$ ⁴, for the MR-SEA II load mass. The presence of the plant non-idealities, and predominantly of the amplitude- and frequency limitation in the actuator inner velocity loop, further increases the minimum value of interaction mass for coupled stability. Given the effect of the limitations of the experimental prototype used to validate this model, only the cases of $K_{des} = 1.25k_s$ and $K_{des} = 1.5k_s$ are considered for the experimental validation.

5.1.2. Experimental validation in the MR-SEA

The optimized control action was implemented in the MR-SEA, using values of desired stiffness K_{des} 25% and 50% higher than the physical stiffness of the spring. The p and z gains from Table 2 corresponding to the lowest-possible

⁴Although a higher range of stable stiffness regulation can be achieved by increasing the value of the force feedback control gain $k_{p,f}$, this case is not considered because the prototype is unstable for higher values of $k_{p,f}$, when $K_{des} < k_s$, due to unmodeled higher-order dynamics

Table 2: Numerical model results - Minimum environment mass for coupled stability

K_{des}	p	z_{best}	$m_{e,MIN}$	$m_{e,MIN}^{sat}$	$m_{e,MIN}^{freq}$	$m_{e,MIN}^q$
	[1/s]	[1/s]	[kg]	[kg]	[kg]	[kg]
ine	500	150	0.12	0.23	0.54	0.54
$2k_s$	250	100	0.23	0.34	0.66	0.67
	100	50	0.58	0.60	0.93	0.94
ine	500	160	0.10	0.17	0.43	0.43
$1.75k_s$	250	105	0.19	0.26	0.53	0.53
	100	52	0.48	0.48	0.75	0.76
ine	500	180	0.07	0.12	0.31	0.31
$1.5k_s$	250	115	0.15	0.19	0.40	0.40
	100	55	0.37	0.37	0.57	0.58
ine	500	220	0.05	0.07	0.20	0.20
$1.25k_s$	250	133	0.10	0.11	0.25	0.25
	100	62	0.26	0.26	0.40	0.40

value of interaction mass for coupled stability, in the worst case, are chosen (i.e. $p = 500$ for both cases and $z = 220, 180$ for $K_{des} = 1.25k_s$ and $K_{des} = 1.5k_s$, respectively). The stability of the system for these gains is proved in the most destabilizing case, i.e. the application of a constant force to the slider, with instantaneous load removal. In this case, the expected behavior is that of a sustained oscillation of the resulting spring-mass system (note that the stiffness compensator makes no attempt at reducing the load mass), but the energy dissipated through the control action (introduced in order to have the desired stability margins) provides the asymptotic convergence of the slider position - see Fig. 11(a). During high-stiffness display, the MR-SEA was commanded a reference position that was either constant $x_{L,des} = 0$ or oscillating $x_L(t) = A \sin(2\pi f_0 t)$ - $A = 10$ mm, $f_0 = 0.5$ Hz, Fig. 11(b). During the experiment, the subject alternated between the application of a continuous, roughly sinusoidal displacement

to the slider and the application of impulsive, impact-like displacement⁵. The stiffness transfer function $K_v(f)$ was estimated using non-parametric system identification, via the Welch method, using the following relation:

$$\hat{K}_v(f) = \frac{P_{yu}(f)}{P_{uu}(f)}, \quad (17)$$

494 with $P_{yu}(f)$ the cross-spectral density between the input (u) and output (y)
 and $P_{uu}(f)$ is the auto-spectral density of the input. To estimate the stiffness
 496 transfer function, the load position error $x_{L,des} - x_L$ was selected as input, and
 the spring force $F_s = k_s(x_L - x_M)$ was selected as output. The estimated trans-
 498 fer function was resampled in a 51-elements, logarithmically spaced frequency
 vector ($[0.01 - 100]$ Hz), and normalized dividing by K_{des} , obtaining $\bar{K}_v(f)$.
 500 Values of coherence higher than 0.8 were obtained up to approximately 6-7 Hz
 for both cases of $x_{L,des}$, and a Bode plot of the estimated normalized transfer
 502 function is shown in Fig. 11(c).

In the low-frequency range the system behaves as a pure spring, with elastic
 504 constant matching the desired value. The error in virtual stiffness approaches is
 higher than 3 dB up to a frequency that corresponds to 5-6 Hz, beyond which
 506 low coherence values are obtained and the transfer function estimate is no longer
 reliable. The phase of the estimated transfer function is negative, confirming
 508 the non-passivity of the system when $K_{des} > k_s$.

6. Discussion and conclusions

510 This paper investigates the range of virtual stiffness values that can be stably
 and accurately displayed by a stiffness-controlled series elastic actuator (SEA),
 512 in which the motor is a non backdrivable velocity source. As previously shown in
 the literature [14], a velocity-sourced SEA can be passively stiffness-controlled
 514 to display a virtual stiffness K_{des} , only if $K_{des} \leq k_s$, with k_s being the physical
 stiffness of the compliant element. In our paper, we start by generalizing this

⁵This was done in the attempt to excite the system also at frequencies higher than the ones possible through manual perturbation at high stiffness

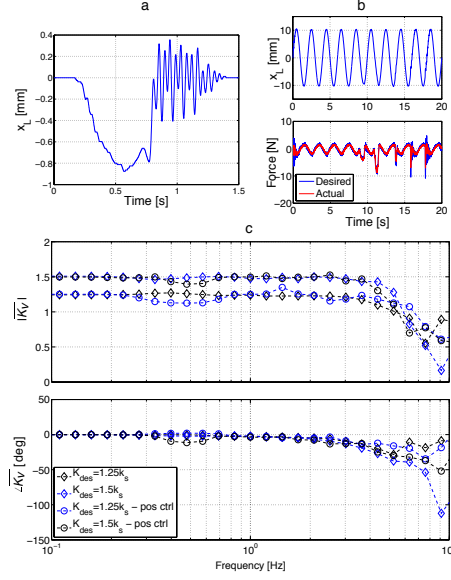


Figure 11: Experimental validation of the controller in the MR-SEA. (a) Application of an increasing load on the stiffness-controlled MR-SEA ($K_{des} = 1.5k_s$) with instantaneous load removal, resulting in decaying oscillations of the slider. (b) Perturbation of the MR-SEA slider when commanded a sinusoidal position tracking ($K_{des} = 1.5k_s$). After the initial transient, the force error is below 1 N before the subject applies perturbation forces to the slider (after 10 s). The desired force is calculated as $-K_{des}x_L$, and is compared with the measured force, i.e. $k_s(x_L - x_M)$. (c) Bode plot of the estimated normalized stiffness transfer functions, for two values of virtual stiffness and for both position tracking cases.

516 finding, demonstrating that, in the context of linear control theory and with
 availability of only position information, no outer-loop force control action can
 518 be applied on the velocity-sourced system to result in a passive system, when a
 stiffness value $K_{des} > k_s$ is desired.

520 We then analyze the effects on coupled stability of a stiffness controller (com-
 manded to impose $K_{des} > k_s$), resulting in a non-passive impedance transfer
 522 function. We show that a frequency-limited lead action in the outer-loop stiff-
 ness controller results in the best accuracy and coupled stability performance,
 524 provided that the system is interacting with a load with inertia higher than a
 threshold m^* . Through our analysis, we derive the analytical expression that

526 provides the minimum environment mass for stable coupled interaction, when
a generic stiffness K_{des} is desired, as a function of the physical stiffness in the
528 actuator and of the controller gains. Our analysis demonstrates that, when
 $K_{des} > k_s$, the value of minimum environment inertia required for a stable in-
530 teraction increases with K_{des} with the scaling factor dependent on the choice of
the controller gains.

532 Using a similar methodology, we address the effect of a non-ideal inner ve-
locity loop on haptic display performance and stability. In particular, we inves-
534 tigate how the minimum value of interaction mass resulting in coupled stability
can be minimized through placement of the stiffness compensator zeros and
536 poles, considering realistic limitations in the bandwidth of velocity control, for
non-backdrivable motors such as those used in [28] for haptic display of MR-
538 compatible robots. We showed that with knowledge of the frequency limitations
of the velocity inner loop (in our case knowledge of the time constant τ of a first-
540 order low pass filter), it is possible to further optimize the stability properties, by
significantly reducing the value of minimum mass for a stable interaction, when
542 the pole or zero at higher frequency is one decade below the cut-off frequency
of the inner velocity loop.

544 We finally validate our model through numerical simulations and experi-
ments in the MR-SEA II, a compliant actuator that includes a non-backdrivable
546 piezoceramic motor as a velocity source, demonstrating capabilities for stable
rendering of virtual stiffness 50% higher than the physical spring stiffness.

548 The main result of our analysis is that coupled stability of a stiffness-
controlled compliant actuator can be obtained when the actuator is controlled
550 to *display* a spring with elastic constant K_{des} , with $K_{des} > k_s$, provided that the
minimum value of mass connected to the series elastic element is higher than a
552 threshold, even in absence of friction. This finding is in contrast with an implicit
assumption that has guided the development of most compliant actuators for
554 human interaction developed so far. Indeed, in most cases [14, 37, 38, 32, 39],
the series elastic element has been designed to be substantially stiff, with the
556 idea of then controlling the actuator to display a lower virtual stiffness. The

choice of “stiff” springs for SEAs poses challenges on the selection of sensors
 558 used to infer interaction force, since they are usually required to have very high
 resolution, in order to minimize quantization errors in the measurement of in-
 560 teraction forces. Consequently, some of the anticipated benefits deriving from
 elimination of a traditional force sensor, in terms of reduction in costs and size,
 562 can be lost in the attempt of incorporating the sensor(s) required to measure
 the deflection of the compliant elements.

564 The results of the presented analysis reveal opportunities for alternative de-
 signs of compliant actuators. Given a range of virtual stiffness values that need
 566 to be implemented through control, it is not necessary to design the compliant
 element so that is stiffer than the stiffest-possible virtual environment for a given
 568 application. Instead, the results of this analysis show that the coupled system
 will be stable if the system is designed so that it includes a minimum amount of
 570 inertia that is connected to the compliant element in all operating conditions.
 Limitations to this approach can be provided by backlash, or in general of non
 572 co-located dynamics, between the series elastic element and the load. In the
 presence of backlash, the value of load mass instantaneously decreases, poten-
 574 tially to a value that is lower than the threshold for coupled stability, until the
 control action responds. The energy introduced in the system during impacts,
 576 coupled with delays in the position measurement and with velocity saturation
 of the plant, is a major source of instability in the experimental implementation
 578 of SEAs.

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